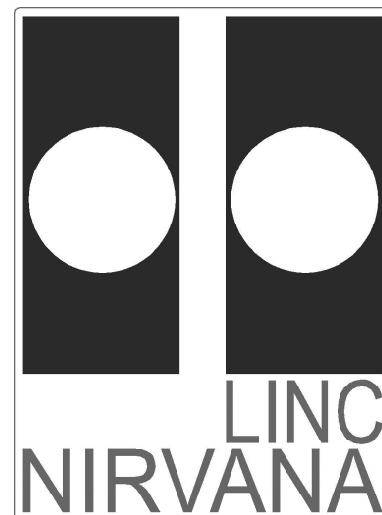


# LINC-NIRVANA

The **L**BT **I**nterferometric **C**amera and  
**N**ear-**I**nfra**R**ed / **V**isible **A**daptive  
**i**nterferometer for **A**stronomy

A collaborative project of the MPIA Heidelberg, INAF-Arcetri,  
Universität zu Köln, and MPIfR Bonn

<http://www.mpia.de/LINC>



## LINC-NIRVANA

### Implementation, Commissioning, and Acceptance: Plans and Procedures

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## Contents

<b>1</b>	<b>Scope</b>	<b>1</b>
<b>2</b>	<b>Applicable Documents</b>	<b>1</b>
<b>3</b>	<b>External Interfaces</b>	<b>1</b>
<b>4</b>	<b>Acronyms and Abbreviations</b>	<b>1</b>
<b>5</b>	<b>Introduction</b>	<b>2</b>
<b>6</b>	<b>Implementation Steps</b>	<b>2</b>
<b>7</b>	<b>Requirements for implementation</b>	<b>3</b>
7.1	Early routine scientific observations . . . . .	3
7.2	LBTO support, logistics, and infrastructure . . . . .	3
7.3	Implementation team . . . . .	4
<b>8</b>	<b>Implementation Plan</b>	<b>5</b>
<b>9</b>	<b>Commissioning Goals</b>	<b>8</b>
<b>10</b>	<b>Commissioning Plan</b>	<b>11</b>
<b>11</b>	<b>Requirements for acceptance</b>	<b>11</b>
<b>12</b>	<b>Acceptance Plan and Procedures</b>	<b>12</b>
12.1	Preliminary Acceptance Laboratory . . . . .	12
12.2	Preliminary Acceptance Mt. Graham . . . . .	12
12.3	Final Acceptance Mt. Graham (optional) . . . . .	12

## List of Figures

1	Implementation schedule . . . . .	7
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## List of Tables

1	Manpower for the implementation/commissioning team . . . . .	4
2	Implementation schedule . . . . .	6
3	Major characterisation/verification items for the commissioning of LINC-NIRVANA . . . . .	9
4	Acceptance milestones for instrument categories . . . . .	11

## 1 Scope

This document presents the plans and procedures for implementation, commissioning, and acceptance of LINC-NIRVANA (LN) at the Large Binocular Telescope (LBT), as far as they have been established at the time of writing.

## 2 Applicable Documents

No.	Title	Number and Issue
1	Scientific and Technical Requirements	LN-MPIA-FDR-GEN-003 1.0
2	Operations and Maintenance Requirements	LN-MPIA-FDR-GEN-004 1.0
3	External Requirements	LN-MPIA-FDR-GEN-005 1.0
4	Assembly, Integration and Testing Plan	LN-MPIA-FDR-AIT-002 1.0
5	Alignment Plan	LN-MPIA-FDR-AIT-003 1.0
6	Management	LN-MPIA-FDR-PM-001 1.0
7	Handling and Shipping	LN-MPIA-FDR-AIT-005 1.0
8	Memorandum of Understanding for the Design, Construction and the Usage Rights of LINC-NIRVANA, an Interferometric Beam Combiner with Multi-Conjugated Adaptive Optics for the Large Binocular Telescope	n.a. 1.2
9	LBT SAC Activities 2000-2004	n.a. 1.0

## 3 External Interfaces

Item	Short description
LBTO	LBT Observatory interactions

## 4 Acronyms and Abbreviations

AIT	Assembly, integration and testing
AO	Adaptive optics
DM	Deformable Mirror
GWS	Ground-layer Wavefront Sensing System
LBT	Large Binocular Telescope
LBTO	LBT Observatory
LN	LINC-NIRVANA
MCAO	Multi-conjugate adaptive optics
MHWS	Mid-high-layer Wavefront Sensing System
MoU	Memorandum of Understanding
n.a.	Not applicable
OPD	Optical path difference (“piston”)
PDR	Preliminary Design Review
SAC	Scientific Advisory Committee

## 5 Introduction

LINC-NIRVANA (LN) will be implemented at the LBT in a step-wise fashion. Each of a total of four implementation steps realizes a distinct operation mode. The final instrument configuration attained after successful completion of all four implementation steps is defined as the baseline configuration of LN in the Memorandum of Understanding (MoU) of the LN consortium partners.

Each implementation step is connected with its own commissioning phase(s) and followed by a period of routine science observations. In this document, the major commissioning goals are described for each implementation step, together with a time schedule, whereas a detailed commissioning plan at task level will be prepared by the time the instrument is shipped to Arizona (cf. LN-MPIA-FDR-AIT-005).

Plans and procedures for instrument acceptance are addressed and described as far as they have been established at the time of writing (spring 2005).

## 6 Implementation Steps

Due to the great complexity of the baseline configuration of LN, the instrument will be implemented in a stepwise fashion.

1. LINC mode:

Diffraction limited near-infrared interferometric imaging with a pair of single on-axis natural guide stars, adaptive optics (AO) sensing systems. The goal here is to obtain scientifically useful data as soon as possible and to aim for a performance level qualified for routine science operation. This implementation level will use a single pyramid and star enlarger in each of the two mid-high layer wavefront sensors (MHWS) to perform AO corrections with the adaptive secondary mirrors of the LBT or (alternatively) with the deformable mirrors (DM) of LN.

2. GWS demonstration:

Demonstration of Ground Level Wavefront Sensors (GWS) driving the adaptive secondaries of the telescope, with a best effort to obtain interferometric measurements.

3. MCAO (single-arm):

Multi-conjugate Adaptive Optics (MCAO) using a single telescope aperture (non-interferometric). This mode uses the GWS and MHWS simultaneously. The goal here is to demonstrate the enhanced field of view and sky coverage afforded by MCAO. Optionally, a three-layer experiment will be performed employing an additional LN DM.

4. NIRVANA mode (MCAO interferometry):

This is the final operating mode of LN, with scientific interferometric observations and MCAO with both GWS (driving the adaptive telescope secondaries) and both MHWS (driving one LN DM per arm).

Even though the stepwise implementation at the telescope is the key to managing the complexity of LN, it is foreseen to ship the complete final hardware and infrastructure (see LN-MPIA-FDR-AIT-005), i.e. the baseline instrument configuration in the MCAO interferometry (NRIVANA) mode. This is for financial and manpower reasons. Therefore, the instrument will be fully assembled, integrated, and tested in the MPIA integration facility (see LN-MPIA-FDR-AIT-002) followed by a final test phase leading to preliminary acceptance Europe.

Afterwards, the instrument will be disassembled to unit level and shipped to Arizona. On Mt. Graham an initial, preparatory phase of (re-)assembly, system integration, and functionality testing is foreseen to take place in a dedicated instrument preparation/integration area on the ground floor of the LBT building. Details of this integration facility are still under discussion with LBTO, but see LN-MPIA-FDR-GEN-005 for requirements from the LN project. The assembled instrument will then be lifted by crane to the instrument platform, followed by a phase of final alignment and integration with the telescope. In the subsequent implementation phase, the instrument may have to be returned to the integration/preparation area for specific tasks, thereby optimizing the time during which the implementation team can access the instrument. This will be coordinated with LBTO.

## 7 Requirements for implementation

### 7.1 Early routine scientific observations

Periods of routine scientific observations are foreseen between the individual implementation/commissioning phases of the instrument.

The LINC mode is therefore considered to be a regular operation mode of the instrument. The other implementation levels must preserve this mode of instrument operation with switch-over from one night to the next.

### 7.2 LBTO support, logistics, and infrastructure

LN imposes a number of requirements on LBTO for support, logistics, and infrastructure. This includes the permanent mounting at the foreseen focal station, the personnel support to be provided by LBTO, the instrument integration/preparation area, services such as ethernet, power, and cooling, access to the control room, requirements on instrument handling, and on accommodation for the implementation/commissioning team and observers.

These high-level requirements towards LBTO are described in LN-MPIA-FDR-GEN-005 (FDR chapter 2) and shall not be repeated here in full detail. However, one particular requirement shall be emphasized here which is the option to bring the instrument back from the telescope platform to the integration/preparation area (see Section 6). As far as realistically possible, the preparation area should be ready for access by LN during the time scheduled for implementation and commissioning.

Table 1: Manpower for the implementation/commissioning team

Phase	Engineers							Astronomers/ physicists	Total
	Optics	Electronics	Detect- ors	Soft- ware	Mech- anics	Cryo- mech- anics	Total Engin- eers		
<b>AIT at LBTO:</b>									
peak activity	2	2	2	2	2	1	11	4	15
routine activity	1	1	1	1	1	1	6	3	9
<b>Commissioning:</b>									
peak activity	1	1	1	1	1	1	6	3	9
routine activity	1	1	1	1			4	3	7
<b>Science:</b>									
early phase	1	1		1			3	2	5
routine observations		1		1			2	1	3

### 7.3 Implementation team

The composition of the LINC-NIRVANA implementation/commissioning team is summarized in Table 1. In the early phases, the various categories of engineers have two people assigned (except cryo-mechanics) in order to permit day and nighttime work. In the later phases, several categories are not represented, but their engineers will provide remote support via internet, VideoCon or phone. Several of the foreseen astronomers have also good knowledge in various engineering aspects or laboratory physics and will therefore provide support already in the early integration phases.

## 8 Implementation Plan

We intend to ask the LBTO for commissioning time for LINC-NIRVANA as follows:

For each of the four implementation steps of LINC-NIRVANA we will request 12 nights within a three-week time interval which will be subdivided as follows:

First week:	1 night arrival+setup plus 6 nights of commissioning	— telescope access not needed — on telescope
Second week:	resting, evaluation of results, and preparations for the remaining week	— commissioning team will leave the mountain
Third week:	1 night arrival+setup plus 6 nights of commissioning	— telescope access not needed — on telescope

The four implementation and commissioning periods are separated by intervals of 3 – 4 months which will be devoted to evaluation and documentation of results at the home institutions of the LINC-NIRVANA partners. Also scheduled between the commissioning phases will be periods of routine science observations. The allocation of these scientific nights will be coordinated by the Italian and German partners in the usual way.

This approach leads to the tentative schedule presented in Table 2 and Fig. 1 for the activities related to implementation, commissioning, and acceptance (see also LN-MPIA-FDR-PM-001, FDR Chapter 14).

This schedule is preliminary and depends not only on progress with LN, but also on the allocation by LBTO of time slots for the planned activities.

Within the allocated nights at the telescope, each run will have a beginning-of-run alignment phase in both daytime and nighttime. In LINC mode, the GWS units will be installed, but will not necessarily need to be aligned. Twice per year, dedicated summer-winter alignment phases are foreseen.

Note that  $4 \times 12$  nights of commissioning of the several LINC-NIRVANA modes will most likely not be sufficient to achieve a performance level similar to that expected from a fully commissioned instrument. This is in line with the strategic character of LINC-NIRVANA within the LBT instrument ensemble. We will certainly have to re-evaluate the needs for further test time at the telescope after the first allocated commissioning runs.

Table 2: Implementation schedule

Phase	Time interval	Activity	Nights*
• Assembly, integration and testing (AIT) at LBTO	2007-01-15 — 2007-03-18	AIT in LBTO integration facility	0
	2007-03-19 — 2007-04-01	AIT on LBT platform	0**
• Implementation Step 1: LINC mode	2007-04-02 — 2007-04-22	Commissioning LINC mode	12
	2007-04-22	Readiness LINC mode	
	2007-04-23 — 2007-09-06	Evaluation/documentation of results 1	0
• Implementation Step 2: GWS demonstration	2007-09-07 — 2007-09-29	Commissioning GWS	12
	2007-09-29	Readiness GWS	
	2007-09-30 — 2007-12-30	Evaluation/documentation of results 2	0
• Implementation Step 3: MCAO (single-arm)	2007-12-31 — 2008-01-21	Commissioning single-arm MCAO	12
	2008-01-21	Readiness single-arm MCAO	
	2008-01-22 — 2008-04-20	Evaluation/documentation of results 3	0
• Implementation Step 4: NIRVANA mode	2008-04-21 — 2008-05-12	Commissioning NIRVANA mode	12
	2008-05-12	Readiness NIRVANA mode	
	2008-05-13 — 2008-08-10	Evaluation/documentation of results 4	0
• Instrument delivery	2008-08-11 — 2008-09-25	Documentation update appropriate to facility instruments	0
	2008-09-26	Acceptance by LBTO for facility instrument	0

\* The number of nights we will request from LBTO for commissioning.

\*\* During initial instrument installation at the LBT platform, it may happen in one or two instances that the telescope will not be ready for nighttime operations.

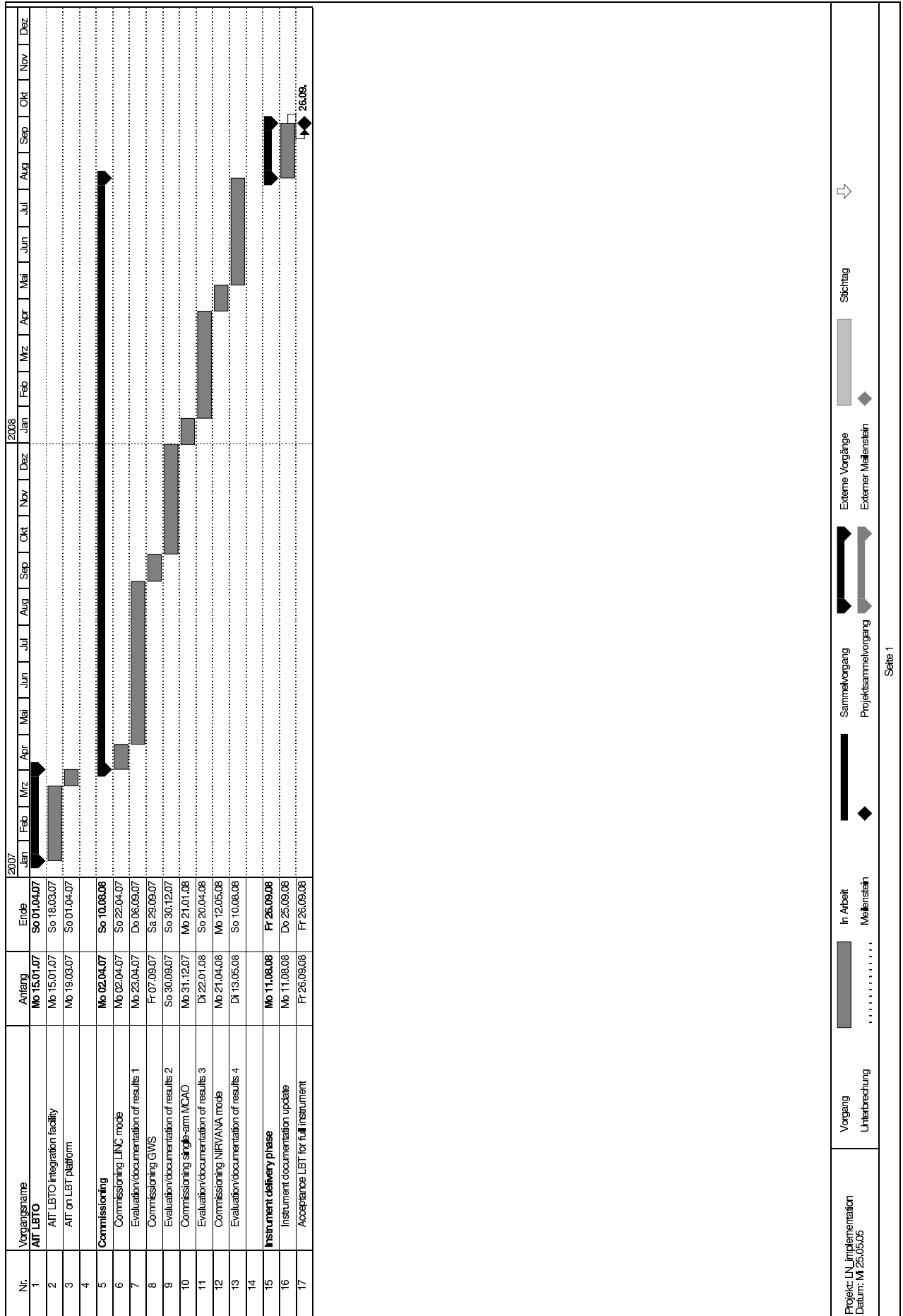


Figure 1: Implementation schedule

## 9 Commissioning Goals

Primary verification items (main drivers) are

- high fringe contrast,
- a field of view (FoV) as large as possible.

In turn these depend on

- Strehl ratio (wavefront error)
- homotheticity (the less fulfilled the smaller the FoV)
- iso-planatic patch size (an-iso-planatism, asterism dependent)
- differential piston ( $OPD < \lambda/10$ , where  $\lambda = 1.13\mu$ ) and iso-piston patch size
- field overlap (from both telescopes)  $<$  PSF
- derotation smearing (max.  $30^\circ$  in parallactic angle) implies alignment of the optical axes of telescope and detector to better than 2 pixels

The principal verification and characterisation goals for the commissioning of the instrument in its four implementation modes are listed in Table 3.

Other characterisation items (on-sky):

- Image scale science detector (specified is 5.11 mas per pixel to achieve 2-pixel sampling at  $1.13\mu$  given single telescope  $f/87.9$  beams, i.e. FoV is  $10.5'' \times 10.5''$  with a  $2048 \times 2048$  pixel detector)
- Characterisation of fringe tracker FoV (specified  $1' \times 1.5'$ ) with respect to iso-piston patch size; done via fringe contrast measurements over the FoV
- Fringe tracking stability (depending on seeing)
- On-sky check of pyramid sensor stability for both GWS and MHWS
- On-sky optimization of pyramid sensor positioning algorithm for both GWS and MHWS
- Differential piston “caused by” AO residuals: (a) spatial variation, (b) temporal variation
- Differences in the images of the two telescopes affecting individual image scales, different distortions, rotations (homotheticity differences)
- On-sky development of a strategy to minimize these differences
- Performance tuning of MCAO and fringe tracking loops
- Astrometric error, asterism dependent
- Photometric error

An attempt will be made to study several of these items as a function of telescope inclination (flexure tests).

Table 3: Major characterisation/verification items for the commissioning of LINC-NIRVANA

Pre-conditions:		(1) Reference wavelength K-band (2) Elevation $\geq 30^\circ$ above horizon (3) Seeing in R-band $\leq 1''$ at employed elevation (4) Suitably selected asterisms (appropriate to instrument modes 1–3)	
Impl. step	Instrument mode	Item	Commissioning goal
1	LINC	<ul style="list-style-type: none"> <li>• <u>Bright reference sources:</u>                Time-averaged on-axis Strehl ratio                Time-averaged on-axis interferometric performance                Overall throughput                Thermal background (accumulated emissivity)                Isopiston patch at site                 Image scale</li> <li>• <u>Faint reference sources:</u>                Fringe contrast as a function of brightness (fringe tracking star)                Limiting magnitude for fringe tracking</li> <li>• <u>Operations:</u>                Fringe tracker usable field                Duration of continuously closed loop (with above performances)                Duration of startup, i.e setup of complete instrument plus first target acquisition                Re-acquisition time                Instrument recovery time (full re-boot and re-acquisition)                Speed of closed-loop nodding up to an amplitude of <math>5''</math></li> </ul>	30% 50% of target routine performance  30% Characterize, isolate contributors, and correct (when possible) Characterize interferometric performance a function of off-axis distance $\approx 5.11$ mas/pixel  Characterize fringe contrast as a function of J-magnitude Determine J-magnitude for 50% fringe contrast degradation; verify pertinent J-magnitude. Goal: J=17 mag  Goal: $30'' \times 30''$ 30 min  $< 1$ hr night time  Goal: 7 min, maximum 20 min $< 1$ hr  Goal: 1 sec/arcsec, max. 5 sec/arcsec
2	GWS demo	<ul style="list-style-type: none"> <li>• <u>Bright reference sources:</u>                Ground-layer correction                Turbulence strength/variance of phase fluctuation above site</li> <li>• <u>Faint reference sources:</u>                Limiting magnitude for ground-layer correction</li> <li>• <u>Operations:</u>                Continuous operation                Re-acquisition time (target and asterism)                Interferometric observations</li> </ul>	FWHM reduction 0.5, minimum 0.7 Characterize the $C_N^2$ profile as a function of height  Determine R-magnitude (integrated over all reference stars) for 50% degradation of Strehl ratio. Goal: R=15  30 min Goal: 10 min, maximum 30 min  Best effort, no requirement

Continuation of Table 3

Impl. step	Instrument mode	Item	Commissioning goal
3	Single-arm	<ul style="list-style-type: none"> <li>• <u>Bright reference sources:</u> Time-averaged Strehl ratio in field Strehl variation over FoV (10.5'' × 10.5'')</li> <li>Isoplanatic patch</li> <li>• <u>Faint reference sources:</u> Limiting magnitude for mid-high layer correction</li> <li>• <u>Operations:</u> Closed-loop operation Re-acquisition time (target and asterism) Setup time to change layer height (movement of DM, MHWS)</li> </ul>	<p>30% 5%</p> <p>Characterization for selected asterisms</p> <p>Determine R-magnitude (integrated over all reference stars) for 50% degradation of Strehl ratio. Goal: R=15</p> <p>30 min Goal: 10 min, maximum 30 min</p> <p>Goal: 1 min, maximum 5 min</p>
4	NIRVANA	<ul style="list-style-type: none"> <li>• <u>Bright reference sources:</u> Same as for LINC mode</li> <li>• <u>Faint reference sources:</u> Limiting magnitude for fringe tracking</li> <li>Limiting magnitude for MCAO</li> <li>• <u>Operations:</u> Fringe tracker usable field Re-acquisition time (target and asterism) Other items: same as LINC mode</li> </ul>	<p>Verify goal of J=17 mag (50% degradation of interferometric performance)</p> <p>Verify goal of R=15 mag for both GWS and MHWS (50% Strehl ratio degradation)</p> <p>Goal: 1' × 1.5'</p> <p>Goal: 10 min, maximum 30 min</p>

Table 4: Acceptance milestones for instrument categories

	Strategic instruments	Facility instruments
Preliminary Acceptance Laboratory	×	×
Preliminary Acceptance Mt. Graham	×	×
Final Acceptance Mt. Graham		×

## 10 Commissioning Plan

For the purpose of this document, the fundamental verification and characterisation items and goals have been identified in the previous section. A more detailed commissioning plan at task list level will be prepared in due time, presumably before preliminary acceptance Europe and at the latest before implementation activities on Mt. Graham commence.

## 11 Requirements for acceptance

The requirements for acceptance by LBT have yet to be established in detail by LBTO. Here, it is assumed that the milestones summarized in Table 4 will have to be passed.

This assumption follows the recommendation that the Scientific Advisory Committee (SAC) made on its meeting on 3–4 May 2002 (LBT SAC Activities 2000-2004, p. 77):

*The Project should establish a “preliminary” and “final” acceptance procedure based on the following general steps:*

1. *Preliminary Acceptance Laboratory — The instrument is deemed safe, useful, and reliable enough to bring to the telescope for commissioning.*
2. *Preliminary Acceptance Mt. Graham — The instrument meets minimum, non-negotiable, performance requirements. These requirements correspond to the performance specifications set out in the procurement contract.*
3. *Final Acceptance Mt. Graham — The instrument meets support, maintainability, and negotiable scientific performance targets. These targets correspond to the performance goals laid out in the contract.*

*The LBT Director should have the authority to grant final acceptance.*

## **12 Acceptance Plan and Procedures**

### **12.1 Preliminary Acceptance Laboratory**

This is for strategic instrument status.

- Instrument fully assembled in lab
- Alignment procedures successfully established, applied as far as they correspond to lab level, and documented
- Fringe tracking on lab level within requirements and documented
- MCAO lab experiment performed and results documented using as much as possible of the original LN hardware
- A list of instrument characteristics determined (this list will be compiled)

### **12.2 Preliminary Acceptance Mt. Graham**

This is for strategic instrument status.

Successful commissioning and documentation of LINC mode, i.e. implementation step 1.

### **12.3 Final Acceptance Mt. Graham (optional)**

This is for facility instrument status. To achieve it is desired after implementation step 4, NIRVANA mode.